FLIGHT TEST SAFETY WORKSHOP

First Flight Into Category III Weather

Presented to: Flight Test Safety Workshop

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References

- FAA Advisory Circular 20-57A
- FAA Order 8400.13D
- C-141 All Weather Landing System (AWLS)
 Flight Test Report: Optimization and Pre Experimental Phases
- C-141 All Weather Landing System Engineering Support



The Program

- Airplane was a <u>C-141A</u> modified to be able to accurately track a Category III Beam
- Air Force program working with the FAA
 Technical Center developing requirements
 and procedures for Category III operations
- Category III ILS was only available at the FAA Technical Center
- Category II beams were usable, but had some limitations

Primary Test Sites

- FAA Technical Center, Atlantic City, NJ
- Wright-Patterson Air Force Base, OH
- Dayton-Cox Municipal Airport, OH
- New Orleans International Airport, LA

ILS Categories for precision instrument approach and landing

Approach category	Decision height or alert height	RVR	Visibility minimum	Notes
l	200 feet	2400 ft	1600 ft	Either visibility not less than 2400 ft or a runway visual range not less than 1,800 ft on runway with touchdown zone and centerline lighting. FAA Order 8400.13D allows for special authorization of CAT I ILS approaches to a decision height of 150 feet with RVR >= 1,400 feet. The aircraft and crew must be approved for CAT II operations and a heads-up display in CAT II or III mode must be used to the decision height. CAT II/III missed approach criteria apply.
II	100 feet	1200 ft	N/A	RVR (1,200 ft) (ICAO and FAA)
IIIa	50 ft < DH < 100 ft	700 ft	N/A	
IIIb	0 < DH < 50 feet	300 ft < RVR < 600 ft	N/A	RVR > 160 ft (ICAO and FAA). This is currently the best in operation.
IIIc	No DH	No RVR	N/A	Requires guidance to taxi in zero visibility.



Historical Perspective

- ILS Developed in 1939
- Blind Landing Experimental Unit (BLEU)
 - Formed in 1945 at RAF Woodbridge
 - First autoland with passengers 1965
- FAA / USAF Interagency Working Agreement signed October 11, 1969
 - Initial flight testing: 1968-69 by Lockheed
 - USAF (First Stage): Dec 13, 1969 Apr 11, 1972
 - USAF (Second Stage): Sep 12, 1973 Feb 20, 1974
 - USAF Validation: Feb 1974 ----



Results

- Lockheed testing was to obtain data to define minimum performance and equipment criteria
- First Stage testing objectives not obtained because of poor performance of the system
- Second Stage required modifications and was a follow-on to Stage 1 for system optimization
- Validation was to conduct operations in actual weather using the optimized system and the procedures developed in the Second Stage

Avionics Modification

- Pilot & Copilot Flight Director Computers
- Sperry AD350 Attitude Director Indicators
- Electronic Attitude Director Indicator (EADI)
- Approach Landing Radar (ALR)
- Maxon Independent Landing Monitor
- ME-1A Synchro Amplifiers
- Automatic Flight Control System mods:
 - Coupler modified for flare optimization
 - Yaw damper
 - Aileron computer



Avionics Modification

- AWLS configured as separate but dual systems
- Excessive lateral and vertical annunciation system
- Runway Distance Remaining (RDR) indicator

Second Stage Objectives

- a) Determine basic flyability (at NAFEC)
- b) Determine effects of beam switching and course shifting on the airborne system
- c) Gather baseline data for CAT III system performance
- d) Establish system reliability and repeatability
- e) Gather data on ILS interference between Cat III aircraft and other large aircraft
- f) Determine abort criteria and procedures
- g) Train and requalify FAA subject pilots

Flight Test Procedures

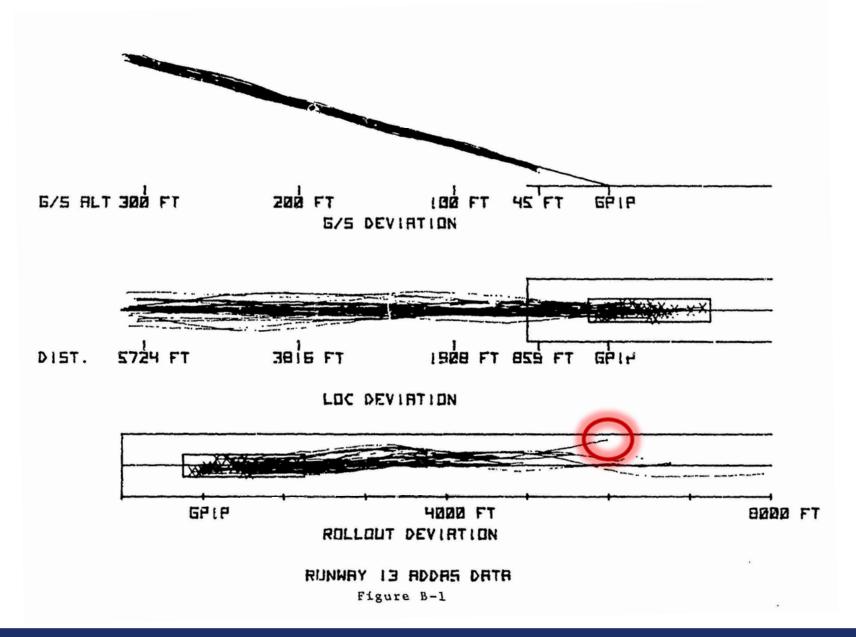
- Standard ILS approaches with qualified test pilots in both seats
- Safety pilot (right seat) callouts:
 - LOC Capture
 - Glideslope Capture
 - Approach Arm Clear (no fault lights after preland test)
 - Land Arm
 - Flare
 - Decrab
 - Rollout

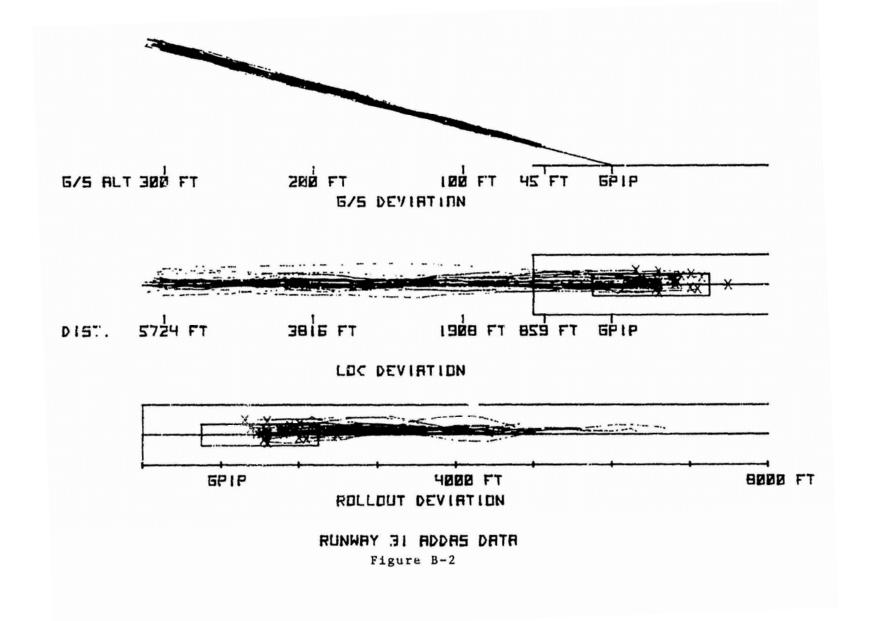
Flight Test Procedures (cont'd)

- Third person familiar with the system in the jump seat
 - Failure in No. 2 system, call "Go-around"
 - Failure in No. 1 system, call "Manual 1, Go-around"
- A pilot-in-the-loop computer simulation and limited flight test / degraded performance analysis completed to ensure safety

Baseline Data

- 51 Approaches at the FAA Technical Center in Sep 1973
- Data Sources:
 - On-board Advanced Digital Data Acquisition System (ADDAS)
 - Theodolite
- Category III ILS on Runway 13; Category II ILS on Runway 31.





Touchdown Box

- Size: ±27'x1500'
- Location:
 - Begins at least 200' from threshold
 - Far end no further down than where the pilot can see at least 4 bars of the 3000' touchdown zone lights

Problems Encountered

- Flare computation did not work properly
- Decrab allowed excessive drift rates
- Autopilot rollout not precise enough

Optimization & Pre-Experimental Phase Recommendations

- Capable of safe flight into Category III
 Weather with restrictions
 - Touch & go only until Runway Distance Remaining display is installed
 - Initially use a decision height to determine lateral rate
 - Prior to zero visibility install lateral rate warning
- INS mode should be used as primary mode for CAT III approaches

Follow-On Testing

- Various facilities visited and briefed in preparation for weather
 - Airport Management
 - Air Traffic personnel
 - Fire Departments
- First flight into Category III weather was a New Orleans in spring of 1974
- Successful program that ushered in Cat III for airline operations

To be remembered

- Expect interruptions and unplanned events
- Establish safety procedures at every airport involved
- Be fastidious about established procedures
 - You can never check the gear too many times!







Type III ILS Facility

- Meets of exceeds ICAO Standards
- Dual frequency localizer that meets CAT III requirement to at least 3000' from the approach end
- Glideslope that meets CAT III requirements to the threshold
- Executive integrity monitors which identify any degradation of signal integrity exceeding CAT III standards

Type III ILS Facility (cont'd)

- Far field monitor to identify critical area incursions or signal variations in the far field which may affect signal integrity
- Backup transmitters
- Backup power
- Typically also includes ancillary equipment:
 - Full runway edge, end and in-pavement lighting
 - TDZ lights
 - Full approach lighting system (ALSF-2)
 - Power changeover requirements for critical lighting

